On periodic solutions associated with the nonlinear feedback loop in the non-dissipative Lorenz model

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Abstract. In this study, we discuss the role of the linear heating term and nonlinear terms associated with a nonlinear feedback loop in the energy cycle of the three-dimensional ($X$–$Y$–$Z$) non-dissipative Lorenz model (3D-NLM), where $(X,Y,Z)$ represent the solutions in the phase space. Using trigonometric functions, we first present the closed-form solution of the nonlinear equation

\[ d^2X/d\tau^2 + (X^2/2)X = 0 \]

without the heating term (i.e., $rX$), (where $\tau$ is a non-dimensional time and $r$ is the normalized Rayleigh number), a solution that has not been previously documented. Since the solution of the simplified 3D-NLM is oscillatory (wave-like) and since the nonlinear term ($X^3$) is associated with the nonlinear feedback loop, here, we suggest that the nonlinear feedback loop may act as a restoring force. When the heating term is considered, the system yields three critical points. A linear analysis suggests that the origin (i.e., the trivial critical point) is a saddle point and that the other two non-trivial critical points are stable. Here, we provide an analysis for three types of solutions that are associated with these critical points. Two of the solutions represent closed curves that travel around one non-trivial critical point or all three critical points. The third type of solution, appearing to connect the stable and unstable manifolds of the saddle point, is called the homoclinic orbit. Using the solution that contains one non-trivial critical point, here, we show that the competing impact of the nonlinear restoring force and the linear (heating) force determines the partitions of the averaged available potential energy from the $Y$ and $Z$ modes. Based on the energy analysis, an energy cycle with four different regimes is identified. The cycle is only half of a “large” cycle, displaying the wing pattern of a glasswinged butterfly. The other half cycle is antisymmetric with respect to the origin. The two types of oscillatory solutions with either a small cycle or a large cycle are orbitally stable. As compared to the oscillatory solutions, the homoclinic orbit is not periodic because it "takes forever" to reach the origin. Two trajectories with starting points near the homoclinic orbit may be diverged because one moves with a small cycle and the other moves with
a large cycle. Therefore, the homoclinic orbit is not orbitally stable. In a future study, dissipation and/or additional nonlinear terms will be included in order to determine how their interactions with the original nonlinear feedback loop and the heating term may change the periodic orbits (as well as homoclinic orbits) to become quasi-periodic orbits and chaotic solutions.

1 Introduction

It has been fifty years since Lorenz published his breakthrough modeling study (Lorenz, 1963) that changed our views on the predictability of weather and climate (Solomon et al., 2007). The model has been extensively investigated by researchers in various fields including earth science, engineering, mathematics, philosophy, and physics (e.g., Gleick, 1987; Sprott, 2003; Jordan and Smith, 2007; Anthes, 2011; Hirsch et al., 2013; Strogatz, 2015). Lorenz’s model with three Fourier modes, which represents the solution to the 2-D Rayleigh–Benard equation (Saltzman, 1962; Lorenz, 1963), is known as the three-dimensional Lorenz model (3DLM). In this paper, we use 3D-NLM to refer to the non-dissipative version of the model that is introduced later in the text.

The scientific community agrees that weather is chaotic, with a finite predictability, and that the source of chaos is nonlinearity. Since the degree of nonlinearity is finite within the 3DLM, the impact of increased nonlinearity on system solutions and/or their stability has been studied using generalized LMs with additional Fourier modes (e.g., Curry, 1978; Curry et al., 1984; Howard and Krishnamurti, 1986; Hermiz et al., 1995; Thiffeault and Horton, 1996; Musielak et al., 2005; Roy and Musielak, 2007a). As compared to the 3DLM, some of the generalized LMs have suggested larger Rayleigh number values (or heating parameters) for the onset of chaos, while others have indicated smaller values. This discrepancy may be attributed to different mode truncations (e.g., Curry et al., 1984; Thiffeault and Horton, 1996; Roy and Musielak, 2007a, b, c; Shen, 2014, 2015, 2016) that lead to different degrees of nonlinearity and different systems whose energy may or may not be conserved (e.g., Roy and Musielak, 2007a).

Among studies using generalized LMs, the pioneering study of Prof. Curry (Curry et al., 1984) suggested that chaotic responses disappeared when sufficient modes are included. Recent studies by Prof. Musielak and his colleagues (Musielak et al., 2005; Roy and Musielak, 2007a, b, c) have examined the transition to chaos and the fractal dimensions of generalized LMs, and have emphasized the importance of proper mode truncation in energy conservation. More recent studies (Shen, 2014, 2015, 2016) have discussed the importance of proper Fourier mode selection in extending the nonlinear feedback loop of the 3DLM. The feedback loop, which consists of the two nonlinear terms of the 3DLM, includes a pair of downscale and upscale transfer processes associated with the Jacobian function of the partial differential equation, discussed in Sect. 2. As previously suggested, the original feedback loop may help stabilize the solution for $1 < r < 24.74$ within the 3DLM. The extended nonlinear feedback loop in high-dimensional LMs can provide
negative nonlinear feedback that produces non-trivial stable critical points when \( r < 42.9 \) within a five-dimensional LM and when \( r < 116.9 \) within a seven-dimensional LM. Based on the above, it has been hypothesized that a system’s stability can be improved further with additional modes that can provide negative nonlinear feedback. While the importance of an increased degree of nonlinearity with more Fourier modes has been discussed in recent studies, the competing role of the nonlinear terms with the linear (heating) term and/or dissipative terms deserves to be examined in order to ascertain the conditions under which nonlinear processes may lead to stable or chaotic solutions.

Roupas (2012) and others have indicated that the 3DLM in the dissipative limit, which is referred to as the 3D-NLM, contains two conserved quantities that represent the conservation of \((KE + PE)\) and \((KE + APE)\), respectively. Here, \(KE\), \(PE\), and \(APE\) are the domain-averaged kinetic energy, potential energy, and available potential energy, respectively. These two quantities, \((KE + PE)\) and \((KE + APE)\), are related to the Nambu Hamiltonians (Nambu, 1973; Nevir and Blender, 1994; Floratos, 2011; Roupas, 2012; Blender and Lucarini, 2013). As a result of conservation properties, the collective impact of the nonlinear feedback loop and the linear (heating) term may effectively act as a “restoring” force. The simplicity of the 3D-NLM enables an examination of how the nonlinear feedback loop and the linear (heating) term work together to produce oscillatory solutions (in the phase space). In this work, we address this issue in conjunction with how the available potential energy is partitioned amongst two different Fourier modes, \(Y\) and \(Z\), where \(Z\) is included in order to enable the nonlinear feedback loop.

The paper is organized as follows: In Sect. 2, we present the governing equations and the 3D-NLM, introduce the nonlinear feedback loop, and derive the energy conservation laws. In Sect. 3, we illustrate the role of the nonlinear feedback loop (with \( r = 0 \)) in acting as a restoring force. With inclusion of the heating term, as well as the nonlinear feedback loop, we present a linear stability analysis for the three critical points and then discuss three types of solutions using analytical and numerical methods. The solutions include two types of oscillatory solutions and the so-called homoclinic orbit (Jordan and Smith, 2007; Sprott, 2003). An energy cycle with four regimes is analyzed based on the tendency of \(KE\) and the partition of \(APE\) at different scales (i.e., either \(Y\) or \(Z\)). Concluding remarks are provided at the end. Appendix A discusses the derivation of the 3D-NLM, and Appendix B presents a closed-form solution to the system with \( r = 0 \) using elliptic functions (e.g., Davis, 1960). Appendix C discusses the equations and initial conditions that are used to obtain the solution of the homeclinic orbit.
2 Governing equations and the non-dissipative Lorenz model

The following governing equations for a 2D \((x,z)\), Boussinesq flow are introduced in order to derive the non-dissipative Lorenz model (3D-NLM) and to calculate its kinetic and potential energy:

\[
\frac{\partial}{\partial t} \nabla^2 \psi = -J(\psi, \nabla^2 \psi) + go\frac{\partial \theta}{\partial x} + \nu \nabla^4 \psi, \tag{1}
\]

\[
\frac{\partial \theta}{\partial t} = -J(\psi, \theta) + \frac{\Delta T}{H} \frac{\partial \psi}{\partial x} + \kappa \nabla^2 \theta, \tag{2}
\]

where \(\psi\) is the streamfunction that yields \(u = -\psi_z\) and \(w = \psi_x\) that represent the horizontal and vertical velocity perturbations, respectively, and \(\theta\) is the temperature perturbation. \(\Delta T\) represents the temperature difference between the bottom and top boundaries. The constants, \(g\), \(\alpha\), \(\nu\), and \(\kappa\) denote the acceleration of gravity, the coefficient of thermal expansion, the kinematic viscosity, and the thermal diffusivity, respectively. The Jacobian of two arbitrary functions is defined as \(J(A,B) = (\partial A/\partial x)(\partial B/\partial z) - (\partial A/\partial z)(\partial B/\partial x)\). The crossout symbol indicates the negligence of a term in the dissipationless limit. Equations (1) and (2), with the dissipative terms, were first used in Saltzman (1962) and Lorenz (1963).

The non-dissipative Lorenz model (3D-NLM) is written as:

\[
\frac{dX}{d\tau} = \sigma Y, \tag{3}
\]

\[
\frac{dY}{d\tau} = -XZ + rX, \tag{4}
\]

\[
\frac{dZ}{d\tau} = XY. \tag{5}
\]

Here, \((X,Y,Z)\) represent the amplitude of the Fourier modes. \(\tau = \kappa(1 + a^2)(\pi/H)^2 t\) is the dimensionless time, \(a\) is the ratio of the vertical scale of the convection cell to its horizontal scale. \(H\) is the domain height, and \(2H/a\) represents the domain width. \(\sigma = \nu/\kappa\) is the Prandtl number, and \(r = R_a/R_c\) is the normalized Rayleigh number or the heating parameter. \(R_a\) is the Rayleigh number, \(R_a = goH^3 \Delta T/\nu \kappa\), and \(R_c\) is the critical value for the free-slip Rayleigh–Benard problem, \(R_c = \pi^4 (1 + a^2)^3/a^2\). The “forcing” terms on the right-hand side of Eqs. (4) and (5) are referred to as the linear force, or the heating term \((rX)\), and the nonlinear force terms \((-XZ\) and \(XY)\). Note that as a result of scale selection in the original Lorenz model, the appearance of \(\sigma Y\) comes from \(g \alpha \partial \theta / \partial x\), which is not associated with the dissipative terms. Appendix A provides detailed derivations.

The 3D-NLM is integrated forward in time using the fourth-order Runge–Kutta scheme. Without a loss of generality, only three different values of the normalized Rayleigh number, \(r = 0, r = 25,\) and \(r = 45\) are used, while keeping other parameters, including \(\sigma = 10\) and \(a = 1/\sqrt{2}\), constant. A dimensionless time interval \((\Delta \tau)\) of 0.01 is used and the total number of time steps is 10 000, giving a total dimensionless time \((\tau)\) of 100. A smaller \(\Delta \tau\) is used to improve accuracy. In this study, unless stated otherwise, the initial conditions (ICs) are as follows:

\[(X,Y,Z) = (0,1,0). \tag{6}\]
These settings were used in order to examine the stability of the 5DLM and 6DLM in Shen (2014, 2015), who also discussed the dependence of the solution on different \( r \) and \( \sigma \). To illustrate the unique features of solutions, a very small time step and/or different ICs are used, including \((X, Y, Z) = (\pm \epsilon, 0, 0), (0, \pm \epsilon, 0), \) and \((2\sqrt{\sigma r}, 0, 2\epsilon)\). The first IC is used for showing solutions with a small cycle, and the second is used for discussing solutions with a big cycle. The third IC is used to perform a numerical simulation for the homoclinic orbit (Jordan and Smith, 2007; Sprott, 2003).

### 2.1 The nonlinear feedback loop and energy conservation laws

Nonlinear terms in the 3D-NLM (and 3DLM) have been shown to result from the Jacobian term, \( J(\psi, \theta) \), in Eq. (2). The nonlinear interaction of two wave modes via the Jacobian term can generate or impact a third wave mode through a downscale (or upscale) transfer process. The subsequent upscale (or downscale) transfer process associated with the third wave mode can provide feedback to the incipient wave mode(s). As illustrated in Shen (2014), \( XY \) and \(-XZ\), respectively, represent the downscale and upscale transfer processes that form a nonlinear feedback loop. When new modes are properly included, the feedback loop can be extended. In the following subsections, we discuss the role of the 3D-NLM nonlinear feedback loop in the energy conservation and partition of available potential energy, which, in turn, helps produce oscillatory solutions.

The domain-averaged kinetic energy (\( KE \)), the potential energy (\( PE \)), and the available potential energy (\( APE \)) are defined as follows (e.g., Thiffeault and Horton, 1996; Blender and Lucarini, 2013; Shen, 2014, 2015):

\[
KE = \frac{1}{2} \int_0^{2H/a} \int_0^H (u^2 + w^2) dz dx, \tag{7}
\]
\[
PE = -\frac{1}{2} \int_0^{2H/a} \int_0^H g\alpha(z\theta) dz dx, \tag{8}
\]
\[
APE = \frac{g\alpha H}{2\Delta T} \int_0^{2H/a} \int_0^H (\theta)^2 dz dx. \tag{9}
\]

Through straightforward derivations, we obtain the following:

\[
KE = \frac{C_o}{2} \bar{X}^2, \tag{10}
\]
\[
PE = -C_o\sigma Z, \tag{11}
\]
\[
APE = -\frac{C_o}{2} \frac{\sigma}{r} (Y^2 + Z^2), \tag{12}
\]

where \( C_o = \pi^2 \kappa^2 \left( \frac{1 + \alpha^2}{\alpha} \right)^3 \). \( APE \) is non-positive (i.e., \( APE \leq 0 \)), since any perturbation reduces the energy transformable to \( KE \).

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where \( C_o = \pi^2 \kappa^2 \left( \frac{1 + \alpha^2}{\alpha} \right)^3 \). \( APE \) is non-positive (i.e., \( APE \leq 0 \)), since any perturbation reduces the energy transformable to \( KE \).
Equations (10) and (11) lead to:
\[
KE + PE = C_o \left( \frac{X^2}{2} - \sigma Z \right) = C_1,
\]
while Eqs. (10) and (12) yield:
\[
KE + APE = \frac{C_o}{2} \left( X^2 - \frac{r}{r} (Y^2 + Z^2) \right) = C_2.
\]

With Eqs. (3)–(5), the time derivative for both Eqs. (13) and (14) is zero, so these two equations describe energy conservation. Both \( C_1 \) and \( C_2 \) are constants and are determined by the initial conditions. Thus, if we express \( Z \) and \( Y \) as functions of \( X \), they are single valued. To facilitate our discussions, the contribution to APE from an individual mode is defined as
\[
APE = -\frac{C_o \sigma I^2}{2r},
\]
where \( I = Y \) or \( Z \); therefore, \( APE = APE_Y + APE_Z \). Note that Eqs. (13) and (14) are related to the two Nambu Hamiltonians, \( C = -X^2/2 + \sigma Z \) and \( H = Y^2/2 + Z^2/2 - rZ \) (Nambu, 1973; Nevir and Blender, 1994; Floratos, 2011; Roupas, 2012; Blender and Lucarini, 2013).

From the initial conditions in Eq. (6), we have \( C_1 = 0 \) and \( C_2/C_o = -\sigma/2r \), the latter of which is \(-0.2\) for \( r = 25 \) and \(-0.11\) for \( r = 45 \). Figure 1 provides the time evolution of the conserved quantities: \((KE + PE)\) and \((KE + APE)\) in Eqs. (13) and (14) from the 3D-NLM. At a larger \( r \) (e.g., \( r = 45 \)), a finer \( \Delta \tau \) is required to improve the numerical accuracy of simulated total energy (Fig. 1c). In this study, to facilitate discussions and unless stated otherwise, either \( C_1 \) or \( C_2 \) is assumed to be zero.

### 3 Discussions

In this section, we discuss the competing role of the nonlinear terms and the linear forcing term in the transient solutions and the energy cycle of the 3D-NLM. From Eqs. (3), (4), and (13), we obtain
\[
\frac{d^2X}{dt^2} + M^2X = 0,
\]
and
\[
M^2 = \frac{X^2}{2} - \left( \sigma r + \frac{C_1}{C_o} \right).
\]

The three terms on the right-hand side of Eq. (15b) represent the impact of nonlinearity, the linear (heating) force, and the initial conditions, respectively. Their competing impacts (i.e., their differences) determine the sign of \( M^2 \), and, thus, the characteristics of the solution. Based on the relative magnitude of the initial state that may lead to \( \sigma r + C_1/C_o \leq 0 \) or \( > 0 \), two types of solutions can be identified (Roupas, 2012). In this study, we focus on the case with \( 0 \leq \sigma r + C_1/C_o \) by using \( C_1 = 0 \) in Sect. 3.2 and \( C_1/C_o = X_o^2/2 \) in Sect. 3.3, where \( X_o \) is the initial value of \( X \). To understand the role of the nonlinear terms (i.e., the nonlinear feedback loop), we begin discussions by solving the solution to the equation that contains no nonlinear terms.
3.1 Solutions of the linear system without the nonlinear feedback loop

By assuming no nonlinear terms, we can begin with two equations: \( dX/d\tau = \sigma Y \) and \( dY/d\tau = rX \) and only one conservation law, as follows:

\[
\text{KE} + \text{APE} = \frac{C_2}{2} \left( X^2 - \frac{\sigma}{r} Y^2 \right) = C_2.
\]  

(16)

This linear case yields \( M^2 = -\sigma r \), and Eq. (15) becomes \( d^2X/d\tau^2 - \sigma r X = 0 \). A local analysis suggests that the origin, the only critical point in the linear system, is a saddle point. After straightforward derivations, the solution is:

\[
X = X_1 e^{-\sqrt{\sigma r} \tau} + X_2 e^{+\sqrt{\sigma r} \tau},
\]  

(17)

where \( X_1 \) and \( X_2 \) are constant coefficients. The origin \((X, Y) = (0, 0)\) is a saddle point, and the trajectory is hyperbolic with solutions exhibiting exponential growth and decay. The initial condition, which determines \( dX/dY \) (= \( \sigma Y/rX \)), can help select the proper mode(s). For example, \((X, Y) = (\sqrt{\sigma/r}, 1)\) only provides the growing mode with \((X_1, X_2) = (0, \sqrt{\sigma/r})\), while \((X, Y) = (\sqrt{\sigma/r}, -1)\) leads to the decaying mode with \((X_1, X_2) = (\sqrt{\sigma/r}, 0)\). The former and latter display the properties of unstable and stable manifolds, respectively (Ide et al., 2002). For the nonlinear case, a “current” state (i.e., ICs) may vary with time. Therefore, either mode may appear at different stages. For example, as shown with the homoclinic orbit in Sect. 3.3, a trajectory with an initial point of \((X, Y, Z) = (2\sqrt{\sigma r}, 0, 2r)\) may approach the origin at a decay rate of \( \sqrt{\sigma r} \), while a trajectory beginning near the origin may depart at a growth rate of \( \sqrt{\sigma r} \). Based on Eqs. (16) and (17), although the time change of \((\text{KE} + \text{APE})\) remains zero, the KE produced using only the linear (heating) force has no upper limit. Such an outcome could violate the linear assumption, and, thus, nonlinearity should be included.

3.2 Nonlinear solutions and the nonlinear restoring force for \( r = 0 \)

Here, we examine the impact of nonlinear terms using a special case with \( r = 0 \) and \((X, Y, Z) = (0, 1, 0)\), leading to \( M^2 = X^2/2 \). Thus, Eq. (15) becomes:

\[
\frac{d^2X}{d\tau^2} + \frac{X}{2}X = 0.
\]  

(18)

As compared to the case with \( r \neq 0 \) in Eqs. (13) and (14), the energy conservation laws obtained using \( r = 0 \) are:

\[
\frac{X^2}{2} - \sigma Z = 0,
\]  

(19)

\[
Y^2 + Z^2 = 1,
\]  

(20)

which, in turn, lead to:

\[
Y^2 + \frac{X^4}{4\sigma^2} = 1.
\]  

(21)
Solutions to the above equation can be obtained as follows:

\[ X^2 = 2\sigma \sin(\phi), \quad (22a) \]
\[ Y = \cos(\phi), \quad (22b) \]
\[ Z = \sin(\phi), \quad (22c) \]

where the phase function, \( \phi \), can be determined from Eqs. (18) and (22a) and is written as:

\[ \phi = \int_{0}^{\tau} X \, d\tau, \quad (23a) \]

which can also be displayed as:

\[ \phi = \int_{0}^{\tau} \int_{0}^{\tau} \sigma Y \, d\tau_1 \, d\tau_2. \quad (23b) \]

To illustrate the solution’s characteristics, Eqs. (22b) and (23b) are solved using the following iterative method:

\[ Y_n = \cos(\phi_n), \quad n = 0, 1, 2 \ldots N \quad (24a) \]
\[ \phi_{n+1} = \int_{0}^{\tau} \int_{0}^{\tau} \sigma Y_n \, d\tau_1 \, d\tau_2, \quad (24b) \]

where \( N \) is the number of iterations. Over a period of time, an initial guess for the phase function is given as \( \phi_0(\tau) = \tau \). We insert the first phase function, \( \phi_0 \), into Eq. (24a) to obtain \( Y_0 \). We then calculate the next phase function, \( \phi_1 \), using \( Y_0 \) and Eq. (24b). The integral in Eq. (24b) is calculated using the trapezoidal rule. We then repeat the above calculations \( N \) times. Numerical results using \( N = 100 \) are provided in Fig. 2. The phase function (Fig. 2a) oscillates with time and varies between 0 and \( \pi \), consistent with Eq. (22a) as a result of \( \sin(\phi) \geq 0 \). Therefore, the solution to Eq. (18) is oscillatory instead of growing or decaying exponentially (as shown in Figs. 2b and c). The nonlinear term in Eq. (18) may be viewed as a nonlinear restoring force. Such a suggestion is consistent with the view (Shen, 2014) that the pair of nonlinear terms \( (-XZ \text{ and } XY) \), leading to the nonlinear term \( (X^2/2) \), can form a nonlinear feedback loop within the 3DLM. When \( \sigma = 10 \) is replaced by \( \sigma = 20 \), an oscillatory solution with a different period is obtained, as shown in Fig. 2d. As a result of the simple method for integral calculation, a fine \( \Delta \tau \) may be required in order to obtain accurate solutions, as indicated by the red and green lines for the results obtained using \( \Delta \tau = 0.0001 \) and 0.01, respectively (see Fig. 2a).

To verify the integral form of the solutions in Eqs. (24a) and (24b), the numerical solutions of the 3D-NLM using \( r = 0 \) (e.g., Eqs. 3–5) are provided in Fig. 3. In panel (a), the blue “dot” indicates the initial temporal evolution of the phase function that is calculated by performing a time integration of \( X \) using Eq. (23a), where \( X \) is obtained from the 3D-NLM; and the red line indicates...
the phase function calculated using Eqs. (24a) and (24b). Both are in good agreement. The simulated trajectories in the $X-Y$ and $X-Z$ sections are elliptic and parabolic (Figs. 3b and c), respectively, consistent with the analytical relationships in Eqs. (21) and (19), respectively. Figure 3d provides the time evolution of oscillatory $Y$ (red) and $X$ (blue), consistent with the results provided in Figs. 2b and c, respectively. As indicated in Appendix B, the oscillatory characteristics of the solution can also be illustrated using elliptic functions.

3.3 Closed form solutions near a non-trivial critical point for $r \neq 0$

In the previous sections, we used Eq. (15) to illustrate the individual impact of the linear (heating) force and the nonlinear feedback loop using cases with $M^2 < 0$ and $M^2 \geq 0$, respectively, neither of which changes the sign during the integration. Here, using the initial condition $(X,Y,Z) = (X_0, 0, 0)$ and $0 < X_0$, we consider a more general case with $M^2 = (X^2/2 - \sigma r - X_c^2/2)$, whose sign may vary during the time integration depending on the relative magnitude of the nonlinearity and the linear (heating) force. $M^2$ has two zeros at $X = \pm X_c$, where $X_c$ is defined as $\sqrt{2\sigma r + X_0^2}$. These are called turning points. Based on an analysis using the WKB approximation (Bender and Orszag, 1978), there appears to be a growing or decaying solution for $|X| < X_c$ and an oscillatory (wave-like) solution for $|X| > X_c$. The former is largely impacted by the linear (heating) force while the latter is impacted by the nonlinear restoring force. Additional analyses are provided below.

To understand the stability of solutions, below, we present a local analysis by linearizing the system with respect to a non-trivial critical point. Using Eqs. (3-5), we first solve for the non-trivial critical point which is $(X_c, Y_c, Z_c) = (X_c, 0, r)$. Here, $X_c$ can be any constant and is selected as the value of the turning point using the justification presented below. From Eqs. (3) and (15), the 3D-NLM system can be written as:

$$\frac{dX}{d\tau} = \sigma Y,$$

$$\frac{dY}{d\tau} = -\frac{M^2}{\sigma} X.$$  

The two equations, with an initial condition of $(X,Y) = (X_0, 0)$, lead to three critical points, $(X_c, Y_c)$ at $(0,0)$ and $(\pm \sqrt{2\sigma r + X_0^2}, 0)$. Note that the non-trivial critical points $\pm X_c$ are turning points due to $M = 0$ in Eq. 15b.

Next, each of the total fields is decomposed into its basic part and perturbation, as follows:

$$V = V_c + V',$$  

where $V$ indicates the total fields $(X,Y,Z)$, $V_c$ represents the basic state obtained from the solution of the critical point, and $V'$ is a perturbation that measures the departure from the critical point. Using the above equation, the linearized system corresponding to the nonlinear system from Eqs. 3-5 is:

$$\frac{ds}{dt} = As,$$  

9
where \( s = (X', Y', Z') \) and the matrix \( A \) that is evaluated at the non-trivial critical point is written as:

\[
A = \begin{pmatrix}
0 & \sigma & 0 \\
0 & 0 & -X_c \\
0 & X_c & 0
\end{pmatrix},
\]

(29)

The above system with Eqs. (28-29) yields eigenvalues of 0 and \( \pm iX_c \), suggesting that the non-trivial critical point \((X_t, 0, r)\) for the linearized system is a stable node as a center. The other non-trivial critical point \((-X_t, 0, r)\) shares the same features. One important feature is the dependence of the solution’s period \( (2\pi/X_c) \) on the initial condition \((X_o, 0, 0)\). Next, to illustrate its periodicity, we present a closed form solution for the nonlinear system.

Equations (13-14), with the initial conditions \((X, Y, Z) = (X_o, 0, 0)\), lead to the following two equations:

\[
\frac{X^2}{2} - \sigma Z = \frac{X_o^2}{2},
\]

(30)

\[
rX^2 - \sigma (Y^2 + Z^2) = rX_o^2.
\]

(31)

Using the same procedures as those provided in Sect. 3.2 (for \( r = 0 \)), the following closed-form solutions (for \( r \neq 0 \)) are obtained:

\[
Y = r\sin(\phi),
\]

(32a)

\[
Z = -r\cos(\phi) + r,
\]

(32b)

\[
X^2 = 2\sigma (r - r\cos(\phi)) + X_o^2,
\]

(32c)

\[
\phi = \int X \, d\tau.
\]

(32d)

Equations (32c) and (32d) are iteratively computed in order to determine the value of \( \phi \) that is used to calculate the solutions of \( X, Y, \) and \( Z \). Note that because \( X_t^2 = 2\sigma r + X_o^2 \), Eqs. (32a) and (32c) lead to:

\[
\left(\frac{X^2 - X_t^2}{2\sigma}\right)^2 + Y^2 = r^2.
\]

(33)

Since \( r \geq r\cos(\phi) \), \( X^2 \) is always positive in Eq. (32c). Without a loss of generality, \( X = +\sqrt{2\sigma (r - r\cos(\phi)) + X_o^2} \) is first discussed, yielding a minimum of \( X_{\text{min}} = X_o \) and a maximum of \( X_{\text{max}} = \sqrt{4\sigma r + X_o^2} \). In a similar manner, given an initial condition of \((X, Y, Z) = (-X_o, 0, 0)\), \( X \) has a maximum of \(-X_o \) and a minimum of \(-\sqrt{4\sigma r + X_o^2} \). Note that the average of \( X_{\text{min}}^2 \) and \( X_{\text{max}}^2 \) is equal to \( 2\sigma r + X_o^2 \), which is equal to \( X_t^2 \).

Applying the initial conditions \((X, Y, Z) = (\sqrt{2\sigma r}, 0, 0)\) that yield \( X_c = X_t = 2\sqrt{\sigma r} \approx 31.62 \), Figure 4 shows the closed form solutions obtained using Eq. (32), as well as numerical solutions of the 3D-NLM obtained using Eqs. (3-5). The results are in good agreement. For any positive \( X_o \), a trajectory beginning with \((X, Y, Z) = (X_o, 0, 0)\) is a closed curve. An oscillatory (closed) trajectory
associated with a center is orbitally stable (Jordan and Smith, 2007). In Sect. 3.4, the time varying energy cycle in Eq. (32) will be analyzed. Next, we discuss the solutions in (32) for the special case of $X_o = 0$.

When considering a solution that begins with and ends at the saddle point, initial conditions of $(X,Y,Z) = (0,0,0)$ are required. At the saddle point, the time derivatives of the 3D-NLM in Eqs. (3-5) are zero. Therefore, numerically, only zero solutions for $X$, $Y$, and $Z$ are possible. Obtaining a non-trivial solution can be attempted by initially adding a small perturbation (i.e., $(X,Y,Z) = (\epsilon,0,0)$) and then performing numerical integrations. A similar approach can be applied in Eq. (32) to obtain an approximate solution. Here, we provide an initial guess for $\phi$, $\phi(\tau) = \tau$, over a target period in Eq. (32d). We then calculate $X$ using Eq. (32c) and $\phi$ using Eq. (32d) through iterations.

Figure 5 displays the solutions determined using Eq. (32). The time evolution of $Y$ is provided in Fig. 5a and suggests that $Y$ increases slowly with time, reaches a maximum at $\tau \approx 2.3$, and then decreases with time to its minimum. The $X$-$Y$ plot in Fig. 5b appears reasonable and is later compared to the analytical solution. In Fig. 5a, the evolution seems "periodic" but displays some irregularities that are different from those obtained using $X_o \neq 0$. This case is further analyzed using the following analytical solution, as well as the numerical solutions in Sect. 3.5.

From Eq. (C3) of Appendix C, the "second" half of the homoclinic orbital solution can be obtained as follows:

$$X(\tau) = \frac{4\sqrt{\sigma}e}{e^{\sqrt{\sigma}\tau} + e^{-\sqrt{\sigma}\tau}}, \quad (34a)$$

$$Y(\tau) = -4\sqrt{\sigma}e \frac{e^{\sqrt{\sigma}\tau} - e^{-\sqrt{\sigma}\tau}}{(e^{\sqrt{\sigma}\tau} + e^{-\sqrt{\sigma}\tau})^2}, \quad (34b)$$

$$Z(\tau) = \frac{X^2(\tau)}{2\sigma}. \quad (34c)$$

When $t \to \infty$, the above solution approaches the origin. Note that a special point on the solution trajectory is $(X,Y,Z) = (2\sqrt{\sigma}r,0,2r)$. Therefore, the "second" half of the solution in Eq. (34) describes the trajectory that begins at $(X,Y,Z) = (2\sqrt{\sigma}r,0,2r)$ and that moves forward in time toward the origin, as shown in lighblue in Fig. 5b. The special point provides an alternative IC for the numerical integration of the 3D-NLM required to simulate (qualitatively) the homoclinic orbit, discussed with numerical results later. When $t \to \infty$, $X(\tau) \approx 4\sqrt{\sigma}te^{-\sqrt{\sigma}\tau}$ and $Y(\tau) \approx -4re^{-\sqrt{\sigma}\tau}$, yielding $X/Y = -\sqrt{\sigma/r}$. These features are the same as those for the linear (stable) solution with a heating term, as outlined in Sect. 3.1. As the system is invariant under $t \to -t$ and $y \to -y$ (Strogatz, 2015), the solution $(X(\tau),-Y(\tau),Z(\tau))$ in backward time, which represents the "first" half of the solution, displays the trajectory that begins at the origin and then moves to the point $(X,Y,Z) = (2\sqrt{\sigma}r,0,2r)$. The total solution that connects the stable and unstable manifolds of the saddle point is called the homoclinic orbit. However, the trajectory "takes forever" to reach the saddle point (as $t \to \infty$), and is, therefore, not a periodic solution. As a result, the "periodicity" of closed-form solutions with $X_o = 0$, as provided in Figure 5, is spurious, suggesting a numerical
error. Unique features of the homoclinic orbit within the analytical solution, as well as the closed-form solution, will be further explored using numerical solutions below.

3.4 An energy cycle of solutions with \( r \neq 0 \)

Here, we analyze the time evolution of energy along the homoclinic orbit and apply it to nearby trajectories with a small energy cycle. In Eq. (36), \( Y^2 \geq 0 \) leads to \( |X| < 2\sqrt{\sigma r} \), which yields a maximum of \( X \) (\( X_m = 2\sqrt{\sigma r} \)) at \( Y = 0 \), denoted by \( Y_m = 0 \). Taking the partial derivative of Eq. (36) with respect to \( X \) indicates that \( Y \) has extrema when \( X = \pm \sqrt{2\sigma r} = \pm X_t \). At \( X = \pm X_t \), \( Z = r \) and \( Y = \pm r \) from Eqs. (13) and (14), respectively, and, thus, \( \text{APE}_Y = \text{APE}_Z \), in other words, equal contributions to \( \text{APE} \) from the \( Y \) and \( Z \) modes. Furthermore, Eq. (36) suggests that \( Y^2 \) initially increases in association with the increase in \( X^2 \) but later decreases in association with an increase in \( X^4 \). The former is consistent with the linear case in Eq. (16), while the latter is consistent with the simplified nonlinear case in Eqs. (19) and (20). The distribution of \( Y \) as a function of \( X \) (i.e., Eq. 36), with the aforementioned four points, is provided in Fig. 6. The energy cycle: 1) begins at point \( A \); 2) goes through \( B, C, \) and \( D \); and 3) returns back to \( A \). The segment from point \( P \) to point \( Q \) is designated as Leg \( P-Q \), where \( P \) and \( Q \) are either one of the following: \( A(X,Y) = (0,0) \), \( B = (X_t,Y_t) \), \( C = (X_m,Y_m) \), or \( D = (X_t,-Y_t) \). Point \( A \) is a saddle point as discussed in Sect. 3.1, (\( X_t,Y_t \) = \( (\sqrt{2\sigma r},r) \)), and (\( X_m,Y_m \) = \( (2\sqrt{\sigma r},0) \)). The energy cycle is referred to as a small (energy) cycle. Here, it should be noted that the homoclinic orbit is not periodic, and that a small energy cycle may appear within the oscillatory trajectories that move close to, but do not pass through, the saddle point. In the discussion provided in the next section, the analysis discussed above is inter-compared to the numerical solutions obtained from the 3D-NLM.

3.5 Numerical solutions with \( r \neq 0 \)

In this section, we first discuss the numerical solutions with small cycles that were analyzed in the previous section, and then present the numerical solutions with large cycles that move around the three critical points. The former are obtained using ICs of \( (X,Y,Z) = (\pm 1,0,0) \), while the latter are simulated with ICs of \( (X,Y,Z) = (0,\pm 1,0) \). Since the features of solutions (e.g., homoclinic orbits) require a high level of numerical accuracy, for Fig. 7, we perform numerical integrations using a very small time step of \( \Delta \tau = 10^{-5} \) but for a shorter period of time (\( \tau = 4 \)). Figure 7a displays oscillatory solutions with small cycles, each of which moves around one of the non-trivial critical points. Figure 7b displays the two oscillatory solutions with a large cycle, moving around all three critical points. The two solutions share the same orbit. The numerical result of the “homoclinic orbit” is shown in Fig. 7c. Fig. 7d shows the evolution of these solutions. While the evolution of energy described by the small or big (energy) cycle will be analyzed later, we first discuss the results of “homoclinic orbit”.
The special features of the "homoclinic orbit" are illustrated using numerical simulations with ICs of \((X,Y,Z) = (2\sqrt{\sigma r},0,2r)\). Here, the "second" half of the orbital solution with \(X > 0\) and \(Y < 0\) is a focus. As indicated by the green line in Figs. 7c-7d, the solution \(Y\) decreases and then increases along the homoclinic orbit. Nearly zero values of \(Y\) in Fig. 7d suggest that the homoclinic trajectory approaches the saddle point and "remains" close to the saddle point for a long period of time, as indicated by a comparison between the homoclinic orbit solution (in green) and the oscillatory solutions (in light blue and red). Between \(\tau = 0\) and \(\tau = 1.7\) in Fig. 7d, the numerical solution in green represents the "second" half of the homoclinic orbit. In Fig. 7c, the corresponding trajectory in the X-Y phase space that appears in the region with positive values of \(X\) compares well with the analytical solution in Fig. 5b. Near the saddle point, \((X,Y,Z)\) are near zero. Therefore, it is easy for numerical errors (e.g., associated with rounding or truncation) to introduce non-zero forcing to drive the flow. When this occurs, the trajectory moves into a region that has negative values of \(X\), and moves around a non-trivial critical point, as shown in Fig. 7c. This move is inconsistent with the analytical solution. However, it is challenging to avoid this type of error since \(\Delta \tau\) is already very small, \(10^{-5}\). Avoiding this type of error may be more complicated in a higher-dimensional space (e.g., the 5DLM), which is beyond the scope of this study.

The above discussion suggests that the homoclinic orbit separates some trajectories that "encircle" a non-trivial critical point from other trajectories that move around the three critical points. Therefore, the homoclinic trajectory that connects the stable and unstable manifolds of the saddle point may be viewed as a separatrix. Two different points that are initially near but not on the homoclinic path will be on periodic orbits that depart from one another, suggesting diverged trajectories. Using the results of Figure 7 as an example, two trajectories, simulated with initial conditions \((X,Y,Z) = (1,0,0)\) and \((0,0,1)\) and shown with red and light blue lines, display "repeated" divergence and convergence. (The two trajectories may give rise to "periodic" divergence and convergence if the ratio of their periods is a rational number.) Therefore, while each of the oscillatory trajectories is orbitally stable, the homoclinic orbit is not. Note that this type of solution dependence on ICs is different from that of a chaotic attractor in dissipative systems (e.g., the 3DLM). Next, we analyze the energy cycle of the oscillatory solutions.

Using the selected ICs, Fig. 8a shows the large cycle that includes two “small cycles” that are anti-symmetric with respect to the saddle point \(A\). The large cycle resembles the wing of a glasswinged butterfly. The right-hand side of the wing displays the same (or comparable) characteristics as those of the cycle in Fig. 6, while the left-hand side wing is antisymmetric with respect to the origin (i.e., the saddle point). While the trajectories in \(X-Y\) and \(X-Z\) are shown in Figs. 8a and 8b, respectively, the distribution of \(\overline{APE_Y}\) and \(\overline{APE_Z}\) as a function of \(X\) is provided in Fig. 8c. Here, \(\overline{APE_Y}\) and \(\overline{APE_Z}\) represent averaged available potential energy from the \(Y\) and \(Z\) modes, respectively. From the perspective of the APE partition, the \(\overline{APE_Y}\) (red) dominates in Leg \(A-B\) (\(D-A\)) while \(\overline{APE_Z}\) dominates in Leg \(B-C\) (\(C-B\)) where nonlinearity is stronger. Alternatively, \(KE\) is predominantly
converted from $\text{APE}_Y$ when $X^2$ is small, and is largely converted from $\text{APE}_Z$ when $X^2$ is large. Therefore, inclusion of the $Z$ mode can lead to the oscillatory solution by enabling the partition of $\text{APE}$ on different scales at different stages (i.e., linear and nonlinear stages). More detailed analysis of the energy cycle is provided later. Here, we briefly discuss the impact of different initial conditions.

In Fig. 8a, a solution may move “around” the saddle point and the two stable critical points, and its trajectory defines the large cycle. With a different initial condition, a solution may go around one of the non-trivial stable points $(\pm \sqrt{2\sigma r + X_o^2}, 0)$, and its trajectory forms a small cycle. For example, Fig. 8d displays results using three different ICs including $X_0 = \sqrt{2\sigma r}$, $\sqrt{2\sigma r} + 15$, and $\sqrt{2\sigma r} + 30$ as shown with black, blue, and red lines, respectively. The three vertical green lines indicate the locations of the corresponding stable critical points at $X_c = \sqrt{2\sigma r + X_o^2}$. Solutions in Figs. 8a and 8d suggest that the larger the value of (initial) $X_o$ is, the further the trajectory may move away from the saddle point. When an initial $X_o$ is distant from zero (i.e., the saddle point), nonlinearity may become dominant as compared to the (linear) heating term. The corresponding solutions (e.g., the blue curve in Fig. 8d) become more symmetric with respect to the green line that passes through the nontrivial stable critical point. As discussed using Fig. 4 in Sect. 3.3, oscillatory solutions with small cycles can be described by the closed-form solutions in Eq. (32).

Figure 9 provides the time evolution of $\text{APE}_Y$ and $\text{APE}_Z$ for the solutions in Figs. 8a-8c. The energy cycle: 1) begins near point $A$, at $A^+ = (0, 0^+) \in$ to be specific; 2) goes through $B$, $C$, and $D$; and 3) returns back to $A$, (i.e., $A^- = (0, 0^-)$). In Leg $A-B$, where the linear (heating) force dominates, the solution $X$ grows gradually with $\{\text{KE}, \text{APE}_Y \downarrow, \text{APE}_Z \downarrow\}$ and $|\text{APE}_Y| > |\text{APE}_Z|$. The upper arrow (↑) and the down arrow (↓) are, respectively, used to indicate an increase and decrease. In Leg $B-C$ (or $C-D$), where the nonlinear restoring force becomes dominant, the solution $X$ increases (or decreases) rapidly with $\{\text{KE}, \text{APE}_Y \downarrow, \text{APE}_Z \downarrow\}$ (or $\{\text{KE}, \text{APE}_Y \uparrow, \text{APE}_Z \downarrow\}$) and $|\text{APE}_Y| < |\text{APE}_Z|$. In Leg $D-A$, solution $X$ decays slowly with $\{\text{KE}, \text{APE}_Y \downarrow, \text{APE}_Z \uparrow\}$ and $|\text{APE}_Y| > |\text{APE}_Z|$. Once the trajectory returns back to point $A^-$, it may experience another small cycle, going to $B^-$, $C^-$, and $D^-$ and returning back to point $A^+$. Here, $B^- = (-X_t, -Y_t)$, $C^- = (-X_m, Y_m)$, and $D^- = (-X_t, Y_t)$. The two “small cycles” form the large cycle, and the time evolution of energy is the same for both wings.

4 Concluding remarks

More than 50 years ago using his forced, dissipative model, Lorenz showed that chaos may appear in the presence of nonlinearity, suggesting that nonlinearity may be the source of chaos. In this study, using the non-dissipative Lorenz model (3D-NLM), we discussed how nonlinearity may act as a restoring force to produce oscillatory solutions. We first presented the closed-form solution of the nonlinear equation $d^2X/dr^2 + (X^2/2)X = 0$, which is derived from the 3D-NLM with $r =$
The corresponding solution is oscillatory (wave-like) and the nonlinear term \((X^2)\) is associated with the nonlinear feedback loop. Therefore, we suggest that the nonlinear feedback loop may act as a restoring force. The closed-form solution obtained using a trigonometric function (Eq. 22a) compares well with the closed-form solution obtained using an elliptic function (Eq. B9); and the simplicity of the former effectively illustrates the fundamental role of nonlinearity in producing oscillatory solutions. While a heating term is considered, a linear analysis shows a saddle point at the origin and two non-trivial stable points at \((X_c, Y_c, Z_c) = (\pm X_t, 0, r)\). Here \(\pm X_t\) depends on the product of \(\sigma\) and \(r\), as well as the initial conditions. Additionally, three types of solutions could be obtained. Two of the solutions display periodic movement around one non-trivial critical point or all three critical points. The corresponding trajectories are orbitally stable. The third type of solution is the so-called homoclinic orbit. Two of the trajectories, beginning at points near the homoclinic orbit, may be diverged, displaying the dependence of the solutions on the initial conditions. The homoclinic orbit is not orbitally stable. Note that this solution dependence is different from that associated with a chaotic attractor in a dissipative system that includes two unstable non-trivial critical points.

The homoclinic orbit connects the stable and unstable manifolds of the saddle point, but it is not a periodic solution. However, the time evolution of energy along the homoclinic orbit can qualitatively depict the energy evolution for trajectories that move around a non-trivial critical point. From the perspective of an energy partition, inclusion of the \(Z\) mode within the 3D-NLM not only introduces additional APE to be transferred to KE, but it also limits KE to be finite, as compared to the linear system. We illustrated that the relative impact of the nonlinear restoring force and the linear (heating) force determines the partition of the averaged available potential energy associated with the \(Y\) and \(Z\) modes, denoted as \(\text{APE}_Y\) and \(\text{APE}_Z\), respectively. Based on the energy analysis, an energy cycle with four different regimes could be identified with the following four points: \(A(X, Y) = (0, 0), B = (X_t, Y_t), C = (X_m, Y_m),\) and \(D = (X_t, -Y_t)\). With the initial perturbation, \((X, Y, Z) = (0, 1, 0), (X_t, Y_t) = (\sqrt{2\sigma}r, r),\) and \((X_m, Y_m) = (2\sqrt{\sigma}r, 0)\). The energy cycle may: 1) begin at (near) point \(A\); 2) go through \(B, C,\) and \(D\); and 3) return back to \(A\). \(|\text{APE}_Y| < |\text{APE}_Z|\) appears when solutions are strongly nonlinear (i.e., in Legs B-C and C-D), suggesting that inclusion of the \(Z\) mode can enable the partition of APE on different scales, leading to nonlinear solutions. Since point \(A\) is a saddle point, the “cycle” is only half of a large cycle, representing one wing of the glasswinged butterfly. A summary of the energy cycle and the large cycle is provided near the end of Sect. 3.

In summary, the nonlinear feedback loop alone may act as a restoring force and the heating term alone can produce a saddle point within the 3D-NLM. The nonlinear feedback loop and the heating term collectively lead to three critical points and three types of solutions. The existence of the homoclinic orbit and two types of oscillatory solutions indicates the importance of the nonlinearity (i.e., the nonlinear feedback loop) and suggests the appearance of diverged trajectories. This type of solution dependence on ICs will be examined and compared with the one associated with a chaotic attractor in the 3DLM and high-dimensional LMs in a future study.
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Appendix A: Derivations of Eq. (3)

In this section, we describe the procedures used to derive Eq. (3). As discussed in Shen (2014a), the nonlinear advection term in Eq. (1) does not explicitly appear within the Lorenz model. Therefore, Eq. (1) can be written as follows:

\[ \frac{\partial}{\partial t} \nabla^2 \psi = g \alpha \frac{\partial \theta}{\partial x}. \] (A1)

Using the \( M_1 - M_3 \) of Shen (2014a), \( \psi \) and \( \theta \) can be represented as:

\[ \psi = C_1 X M_1, \] (A2)

\[ \theta = C_2 (Y M_2 - Z M_3). \] (A3)

Here, \( C_1 \) and \( C_2 \) are defined in Eq. (9) of Shen (2014a). Next, the left- and right-hand side terms of Eq. A1 are expressed using \( M_1 - M_3 \), respectively, as follows:

\[ \frac{\partial}{\partial t} \nabla^2 \psi = -C_1 \left( l^2 + m^2 \right) \frac{dX}{dt} M_1, \] (A4)

\[ g \alpha \frac{\partial \theta}{\partial x} = C_2 Y g \alpha \left( -\sqrt{2} \right) l \sin(lx) \sin(mz) = -C_2 Y g \alpha l M_1. \] (A5)

Here, \( l \) and \( m \), as defined in Shen (2014), represent the horizontal and vertical wavenumbers, respectively. Substituting Eqs. A4 and A5 into Eq. A1, we have:

\[ \frac{dX}{dt} = \frac{g \alpha l}{l^2 + m^2} \frac{C_2}{C_1} Y. \] (A6)

We define a time scale \( T_o \) as:

\[ T_o = (1 + a^2) \frac{\pi^2}{H^2}, \] (A7)

and obtain:

\[ l^2 + m^2 = (1 + a^2) \frac{\pi^2}{H^2} = T_o. \] (A8)

The term \( H \) represents the domain height, and \( a \) is the ratio of the vertical scale of the convection cell to its horizontal scale, (i.e., \( a = l/m \)). With Eq (9) in Shen (2014a), the right-hand side of Eq. A6 becomes:

\[ \frac{g \alpha l}{l^2 + m^2} \frac{C_2}{C_1} Y = \frac{H^2}{\pi^2(1 + a^2)} \frac{g o \pi a}{H} \kappa(1 + a^2) \frac{\Delta T}{\pi} R_e \frac{\alpha}{R_\alpha} Y = \nu(1 + a^2) \frac{\pi^2}{H^2} = \nu T_o Y. \] (A9)
With Eq. A9, Eq. A6 becomes:

\[ \frac{dX}{dt} = \nu T_o Y. \]  \hspace{1cm} (A10)

The above equation then becomes Eq. (3), \( \frac{dX}{d\tau} = \sigma Y \), when \( \tau = \kappa T_o t \) is assumed.

**Appendix B: A closed-form solution for \( r = 0 \) using elliptic functions**

Nonlinear dynamics have been widely studied using solutions to the nonlinear Duffing equation. The Duffing equation that governs certain damped and driven (i.e., dissipative and forced) oscillatory motions (Bender and Orszag, 1978; Jordan and Smith, 2007; Wikipedia) is given by:

\[ \frac{d^2 X}{d\tau^2} + \delta \frac{dX}{d\tau} + \alpha X + \beta X^3 = \gamma \cos(\omega \tau), \]  \hspace{1cm} (B1)

where \( \delta, \alpha, \beta, \gamma, \) and \( \omega \) are constants. When \( \delta = \gamma = 0 \) and \( \alpha = -\sigma r \) and \( \beta = 1/2 \), Eq. B1 appears in the form of the 3D-NLM (Eq. 15). While an exact solution to the general form of the Duffing equation may not have been determined, closed-form, approximated, or numerical solutions have been documented in the literature. In this section, we discuss how closed-form solutions to the special case of the non-dissipative and unforced Duffing equation (i.e., \( \delta = \alpha = \gamma = 0 \) and \( \beta = 1/2 \)) can be obtained and expressed in terms of elliptic functions (Davis, 1960).

The incomplete elliptic integral of the first kind, \( F(\phi, k) \), is written as:

\[ u = F(\phi, k) = \int_0^\phi \frac{d\theta}{\sqrt{1 - k^2 \sin^2 \theta}}, \hspace{1cm} 0 < k^2 < 1. \]  \hspace{1cm} (B2)

The elliptic functions of the Jacobi are defined as the inverse of the elliptic integral, as follows:

\[ sn(u, k) = \sin \phi, \hspace{1cm} cn(u, k) = \cos \phi. \]  \hspace{1cm} (B3)

The elliptic functions have the following properties:

\[ sn(0) = 0, \hspace{1cm} cn(0) = 1, \hspace{1cm} sn^2 u + cn^2 u = 1, \]  \hspace{1cm} (B4)

and their derivatives can be obtained as follows:

\[ \frac{d^2}{du^2} sn(u) = 2k^2 sn^3(u) - (1 + k^2) sn(u), \]  \hspace{1cm} (B5)

and

\[ \frac{d^2}{du^2} cn(u) = (2k^2 - 1)cn(u) - 2k^2 cn^3(u). \]  \hspace{1cm} (B6)

Equations B4 and B5 can be helpful for obtaining a solution to the equation \( \frac{d^2 X}{d\tau^2} + X^3/2 = 0 \) (i.e., Eq. 18). Using Eq. B5 and \( k^2 = -1 \), the solution can be written as:

\[ X = D_1 sn\left( \frac{D_1}{2} \tau + \theta_1, k^2 = -1 \right). \]  \hspace{1cm} (B7)

Through Eq. B6 and \( k^2 = 1/2 \), the solution becomes:

\[ X = D_2 cn\left( \frac{D_2}{\sqrt{2}} \tau + \theta_2, k^2 = 1/2 \right). \]  \hspace{1cm} (B8)
Here, \( D_1, D_2, \theta_1, \) or \( \theta_2 \) are constants and are determined from the initial conditions. \( \theta_1 \) or \( \theta_2 \) can be expressed in terms of complete elliptic integrals of the first kind, \( K(k) = F(\pi/2, k^2 = 1/2) \approx 1.8541 \), which is calculated using Matlab (http://www.mathworks.com/help/matlab/ref/ellipke.html). Additionally, as a result of the definition of the elliptic integral (e.g., Eq. B2) and/or validity of the numerical algorithms for the elliptic functions, \( k^2 \) may not be equal to \(-1\). Therefore, only Eq. B8 is discussed below. In this study, the IC with \( X(0) = 0 \) and \( Y(0) = 1 \) (i.e., \( \frac{d}{d\tau} X(0) = \sigma \)) leads to the following solution with \( \theta_2 = 3K \) and \( D_2 = \sqrt{2\sigma} \):

\[
X = \sqrt{2\sigma} \text{cn}(\sqrt{\sigma \tau} + 3K, k^2 = 1/2). \tag{B9}
\]

Figure B1 displays solutions obtained from Eq. 22a (Fig. B1-a) and Eq. B9 (Fig. B1-b) and their differences (Fig. B1-d). The solutions are very close. This comparison confirms the validity of the method for obtaining closed-form solutions for the 3D-NLM with \( r = 0 \) using elementary trigonometric functions (Fig. 2) that are used to illustrate the role of nonlinearity in producing oscillatory solutions. For the case of \( \alpha \neq 0 \) (i.e., \( r \neq 0 \)), similar procedures can be applied using Eq. B5 or B6 in order to obtain closed-form solutions. The procedures are beyond the scope of the present study.

**Appendix C: Equations for the solution of the homoclinic orbit**

Here, we present equations that are used for obtaining an analytical solution that begins and ends at the saddle point. Equations (13-14) with the ICs \((X, Y, Z) = (0, 0, 0)\) become:

\[
\frac{X^2}{2} - \sigma Z = 0, \tag{C1}
\]

\[
X^2 - \frac{\sigma}{r} (Y^2 + Z^2) = 0, \tag{C2}
\]

which lead to:

\[
Y^2 = \frac{1}{\sigma^2} \left( \sigma r X^2 - \frac{X^4}{4} \right), \tag{C3}
\]

Using the above with Eq. (3) yields:

\[
\frac{dX}{d\tau} = \pm \sqrt{\sigma r X^2 - \frac{X^4}{4}}. \tag{C4}
\]

The term \( X^4 \) in Eq. (C4) can be shown to be equivalent to the term \( X^3 \) in Eq. (15) that is associated with the nonlinear feedback loop. Without this term, the system (in Eq. C4 or 15) becomes linear.

Applying the method described in Chapter 3 of Jordan and Smith (2007), we obtain an analytical solution to the nonlinear system in Eq. (C3) that begins and ends at the saddle point \((0,0,0)\), which are shown in Eqs. (34a-34c).
References

Anthes, R.: Turning the tables on chaos: is the atmosphere more predictable than we assume?, UCAR Magazine, spring/summer, available at: https://www2.ucar.edu/atmosnews/opinion/turning-tables-chaos-atmosphere-more-predictable-we-assume-0 (last access: 6 May 2011), 2011.


Figure B1: A comparison of the nonlinear oscillatory solutions using \( r = 0 \) in Eqs. (22a) and (B9), expressed using an elementary trigonometric function and an elliptic function, respectively. (a) The solution of Eq. (22a) in red. (b) The solution of Eq. (B9) in blue. (c) Both of the solutions in (a) and (b). (d) Differences of the two solutions.
Figure 1: The time evolution of (KE+PE) and (KE+APE) from the 3D-NLM and 5D-NLM. Panels (a) and (b) are provided for $r = 25$ and $r = 45$, respectively. Results obtained from the 5D-NLM are presented in order to show the dependence of solutions on the spatial resolutions. In comparison with panel (b), panel (c) shows the dependence of solutions on the temporal resolution using $\triangle \tau = 0.0001$. All of the fields are normalized using the constant $C_o (=\pi^2 \kappa^2 \frac{(1+a^2)}{A})$. 

Figure 2: Closed-form solutions obtained from Eqs. 24ab using $r = 0$ and the iterative method. (a) The phase function $\phi$ with $\Delta \tau = 0.01$ (green) and $\Delta \tau = 0.0001$ (red) (Eq. 24b). (b) $Y = \cos(\phi)$ (Eq. 22b). (c) $X^2 = 2\sigma \sin(\phi)$ (Eq. 22a). (d) The phase function using a different value of $\sigma (=20)$. 
Figure 3: Numerical solutions obtained from the 3D-NLM (Eqs. 3-5) using r=0. (a) A comparison of the phase functions that are calculated from numerical solutions of the 3D-NLM (blue dot) and closed-form solutions of Eqs. 24ab (red). (b) X-Y plot. (c) X-Z plot. (d) The time evolution of X (blue) and Y (red).
Figure 4: Closed-form solutions obtained from Eqs. 32cd using $r = 25$ and $X\sigma = \sqrt{2\pi r}$ and the iterative method. (a) A comparison of the closed-form solution ($r\sin(\phi)$) from Eq. 32b and the numerical solution of $Y$ from the 3D-NLM. (b) X-Y plot. Closed-form solutions are displayed using black lines and numerical solutions obtained from the 3D-NLM are shown with blue dots. The green line passes through the corresponding critical points at $(X_c, Y_c) = (\sqrt{2\sigma r + X^2_\sigma}, 0)$. 
Figure 5: Closed-form solutions with $X_o = 0$ and the analytical solution of the homoclinic orbit. (a) The closed-form solution $(r \sin(\phi))$ of Eq. 32b. (b) The X-Y plot for the closed-form solutions of Eq. (32) (black) and the analytical solutions of the homoclinic orbit using Eq. (38) (lightblue). The green line passes through the corresponding critical points at $(X_c, Y_c) = (\sqrt{2\sigma r + X_o^2}, 0)$.
A Energy Cycle in the 3D–NLM (analytical, r=25)

Figure 6: An energy cycle in the 3D non-dissipative Lorenz model (3D-NLM), as shown in the X-Y plot. The four points for the selected ICs are identified, as follows: A(X,Y)=(0,0), B=(\sqrt{2}\sigma r, r), C=(2\sqrt{\sigma}r, 0), and D=(\sqrt{2}\sigma r, -r). (\sigma, r)=(10, 25). The energy cycle: 1) begins near the saddle point A, at A=(0^+, 0) to be specific; 2) goes through B, C, and D; and 3) returns near A. KE increases as APE decreases along the upper curve (Legs A-B and B-C), and KE decreases as APE increases along the lower curve (Legs C-D and D-A). From a perspective of potential energy partitioning, |APE_Y| \geq |APE_Z| in Legs A-B and D-A where the linear force dominates, and |APE_Y| \leq |APE_Z| in Legs B-C and C-D where the nonlinear restoring force dominates.
Figure 7: Three types of solutions associated with different ICs from the 3D-NLM (Eqs. 3-5) with $\Delta \tau = 10^{-5}$ and $\tau = 4$. (a) Oscillatory solutions with small cycles. (b) Oscillatory solutions with large cycles. (c) A numerical solution for the homoclinic orbit. (d) The time evolution of the simulated homoclinic orbit (green), and oscillatory solutions with a small cycle (red) and a large cycle (lightblue). Panels (c-d) indicate that the homoclinic trajectory begins at $(X,Y,Z) = (2\sqrt{\sigma r},0,2r)$, approaches the origin, and remains near the saddle point for a much longer period of time as compared to the oscillatory solutions. Note that as a result of accumulated numerical error, the "simulated orbit" travels around the saddle point, moves to the other side in panel (c), and is no longer the original homoclinic orbit.
Figure 8: Solutions obtained from the 3D-NLM (Eqs. 3-5). (a) X-Y plot. (b) X-Z plot. (c) X-APE plot. The black, red, and blue lines show normalized $-KE$, $APE_Y$, and $APE_Z$, respectively. Green lines are plotted at $X = \pm X_t = \pm \sqrt{2} \sigma r$ where $Y^2 = Z^2$. (d) The X-Y plot using different initial conditions, $(X, Y, Z) = (X_0, 0, 0)$. The black, red, and blue curves represent results using $X_0 = \sqrt{2} \sigma r$, $\sqrt{2} \sigma r + 15$, and $\sqrt{2} \sigma r + 30$, respectively. The three green lines pass through the corresponding critical points at $(X_c, Y_c) = (\sqrt{2} \sigma r + X_0^2, 0)$. 
Figure 9: Time evolution for the KE and APE in the 3D-NLM (Eqs. 13-14). Black open circles display $KE$. Red, blue, and green lines indicate $-APE_Y$, $-APE_Z$, and $KE+APE$, respectively. All of the fields are normalized by $C_0$. The gray line is plotted at a value of $\sigma r^2/2$, which is equal to $-APE_Y/C_0$ and $-APE_Z/C_0$ at $X = X_t$. These panels show that the solutions are oscillatory and periodic.